

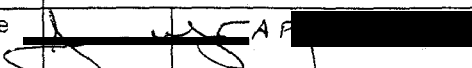

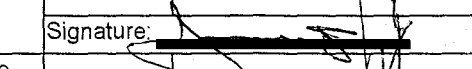
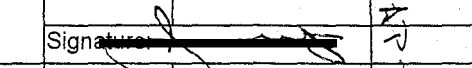
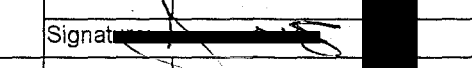
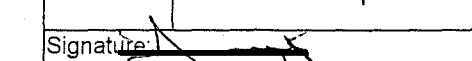
Report Produced By: C.R.AVIATION, INC. Address: 14592 S.W 129th. St Phone: (305) 255-4222
 Tamiami Airport Miami, FL 33186


Airworthiness Directive Compliance Record

Profile ID: **N75908**

Date: 02/01/1996

AD Number Ctry - AD# Rev - Para	1. Subject 2. Comments	Complied 1. Date 2. Time	1. Status 2. Method of Compliance/ Applicability	One or Recur	Next Due 1. Date 2. Time	1. Authorized by 2. Facility 3. Cert. Type 4. Cert. Number
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<u>Manufacturer</u>	<u>Appliance Category</u>	<u>Model</u>			<u>Part #:</u>
Bendix Corporation	Magnetos	D-2000 SERIES			<u>Serial #:</u>
AD 82-20-01 6/14/83	To prevent failure of impulse coupling due to improperly heat treated (soft) flyweights re	11/24/82 2/10/96	P/C/W @ overhaul	One	
AD 94-06-09	capacitor replaced	Nov. 15.97	N/A per model ; Serial #		
<u>Manufacturer</u>	<u>Appliance Category</u>	<u>Model</u>			<u>Part #:</u>
Marvel-Schebler	Carburetors	MA-4			<u>Serial #:</u>
AD 63-22-03 12/2/63	Erratic engine operation or complete engine stoppage	2/10/96	N/A per S/O #	One	
AD 69-24-03 11/29/69	To prevent possible power losses due to blockage of carburetor metering passages by thread	2/10/96	N/A per S/O #	One	
AD 72-06-05 R2 7/3/86	To present looseness or separation of the throttle arm	2/10/96	N/A per S/O #	One	
AD 89-04-02 3/8/89	To prevent possible jamming of the carburetor throttle	2/10/96	P/C/W @ ahl	One	

AD 92-15-16 L 10/5/92	Superseded by 93-19-04	2/10/96	Superseded	One		
					Signature: 	

Propeller



AD 97-18-02

identical prop
for HC () (2.3) (x.v) () ()
type blade
New 30.97 w/A per model #


ATP 

AD 95-11-08

identical Prop
for HC 92WCC
w/A
New 30.97 per model # of
prop.


ATP 

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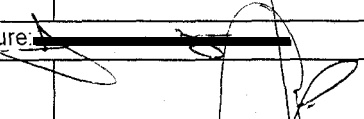
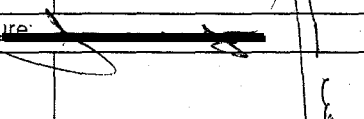
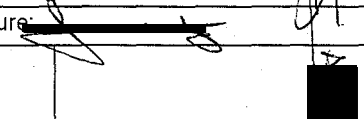
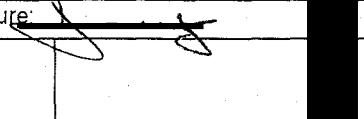
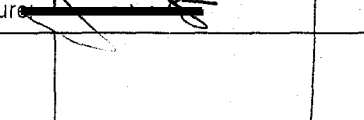

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Manufacturer	Appliance Category	Model	Part #:
Marvel-Schebler	Carburetors	MA-4	Serial #:
AD 93-18-03 10/29/93 <i>187-c16</i>	To prevent disruption of fuel flow to the engine resulting in engine power loss, or engine	2/13/96 <i>2080.8</i> <i>2113.16</i> <i>2210.1</i> <i>Inspected visually - 2 piece</i> <i>c/w per inspect</i> <i>c/w per inspect.</i>	Recur <i>2308.3</i> <i>2407.</i> <i>2506.1</i> <i>2605.8</i> <i>2705.1</i> <i>c/w per inspection</i> <i>c/w per inspection</i> <i>c/w per inspection</i> <i>c/w per inspection</i> <i>c/w per inspection</i> Signature: <i>[Signature]</i>
AD 93-19-04 C 10/18/93	To prevent a disruption of fuel flow to the engine, resulting in engine power loss, engine	2/10/96 <i>P/c/w @ 100h</i>	One Signature: <i>[Signature]</i>
Manufacturer	Appliance Category	Model	Part #:
Airborne Manufacturing	Vacuum Pumps	211CC	Serial #:
AD 79-13-08 6/7/79	To prevent catastrophic failure of the pump and subsequent loss of the vacuum system	2/10/96 <i>P/c/w by replacement</i>	One Signature: <i>[Signature]</i>
AD 93-18-03 10/29/93	Carburetor Venturi	2804.3 2902.6 3001.7 3100.7 3197.81 <i>c/w per inspect. for</i> <i>c/w per inspection</i> <i>c/w per inspection</i> <i>c/w per inspection</i> <i>c/w per inspection</i>	Recur Signature: <i>[Signature]</i>
AD 98-01-06	Carb. Venturi	3297.81 <i>will c/w @ next annual or before</i> <i>@ next annual</i> <i>or 100h.</i> <i>AD note</i>	Signature: <i>[Signature]</i>

AD 95-03-02 *Foracker Air filter* *c/w replaced w/ new*
Nov 15, 97 *asket no problem found* *[Signature]*

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Date: 02/01/1996						
<u>Manufacturer</u> Cessna Aircraft Company		<u>Model</u> 172N		<u>Part #</u> <u>Serial #</u> : 17628052		
AD Number Ctry - AD# Rev - Para	1. Subject 2. Comments	Complied 1. Date 2. Time	1. Status 2. Method of Compliance/ Applicability	One or Recur	Next Due 1. Date 2. Time	1. Authorized by 2. Facility 3. Cert. Type 4. Cert. Number
AD 69-15-03 8/20/69	To detect cracks in the muffler assembly	2/10/96	N/A per 3/10 1/2 year	Recur		Signature: 
AD 71-18-01 8/31/71	To provide correct information concerning fuel tank capacities	2/10/96	N/A per 3/10 1/2 year	One		Signature: 
AD 71-22-02 11/9/71	To decrease the possibility of failure of the nose gear structure	2/10/96	N/A per 3/10 1/2 year	Recur		Signature: 
AD 73-17-01 8/16/73	To advise the pilot concerning proper fuel transfer pump operation	2/10/96	N/A per 3/10 1/2 year	One		Signature: 
AD 74-04-01 2/12/74	To detect cracks in the aft fuselage bulkhead assembly and provide means for repair of cra	2/10/96	N/A per 3/10 1/2 year	One		Signature: 
AD 74-06-02 3/18/74	To prevent possible leakage of carbon monoxide into the cabin heater system	2/10/96	N/A per 3/10 1/2 year	Recur		Signature: 

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Date: 02/01/1996					
<u>Manufacturer</u> Cessna Aircraft Company		<u>Model</u> 172N		<u>Part #:</u> <u>Serial #:</u> 17628052	
AD Number Ctry - AD# Rev - Para	1. Subject 2. Comments	Complied 1. Date 2. Time	1. Status 2. Method of Compliance/ Applicability	One or Recur	Next Due 1. Date 2. Time
AD 76-21-06 10/26/76	To prevent loss of engine oil	2/10/96	N/A per S/W #; model #	One	
AD 77-02-09 2/3/77	Wing flap actuator date code stamp	2/10/96	P/C/W w/ ball nut kit	One	Signature: [Signature]
AD 77-12-08 6/27/77	To prevent unwanted propeller rotation	11/15/77 2/10/96	P/C/W et al	One	Signature: [Signature]
AD 78-26-09 1/1/78	Superseded by 79-10-14	2/10/96	Superseded by 79-10-14	One	Signature: [Signature]
AD 79-08-03 6/6/79	To prevent an inflight electrical system failure, smoke in the cockpit, and/or fire in the	2/10/96	N/A per S/W # ref. cigarette lighter	One	Signature: [Signature]
AD 79-10-14 R1 5/30/88	To provide an alternate source of fuel tank venting in case of fuel tank vent obstruction	2/10/96	N/A per S/W # ref. fuel vent	One	Signature: [Signature]

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Cessna Aircraft Company	172N	<u>Serial #:</u> 17628052

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AD 80-04-08 2/16/80	To preclude the possibility of a fuel leak or an in-flight fire due to contact between a m	2/10/96	P/c/w per S.E. ref. fuel lines	One		
AD 80-06-03 4/21/80	To assure continued structural integrity of the wing flap direct cable, thereby preventing	11/13/80 2/10/96	P/c/w per ref. wing flap follow-up per S.E. 79-16A	One		Signature: [Signature]
AD 81-04-04 1/1/81	Superseded by 81-16-09			One		Signature: [Signature]
AD 81-05-01 3/2/81	To reduce the possibility of fuel depletion due to incorrect fuel quantity markings	2/10/96	N/A per A/c r/v #	One		Signature: [Signature]
AD 81-16-09 8/13/81	To ensure the integrity of the elevator control system	2/10/96	P/c/w ct ok cleared & lubed	One		Signature: [Signature]
AD 82-07-02 4/8/82	To reduce the possibility of rapid loss of engine oil caused by ice obstruction of the eng	11/10/82	P/c/w	One		Signature: [Signature]

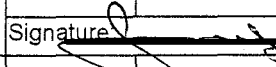
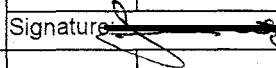
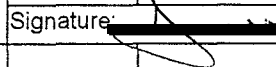
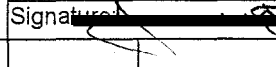
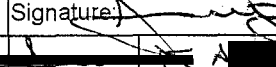
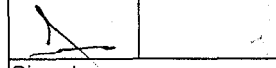
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Cessna Aircraft Company	172N	<u>Serial #:</u> 17628052

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AD 83-10-03 6/30/83	To prevent possible jamming of the elevator control	11/2/83	pk/w ref. control wheel yoke guide	One		Signature: 
AD 83-17-06 9/1/83	To prevent possible destructive aileron flutter	2/10/96	N/A no Robertson STOL kit installed	One		Signature: 
AD 83-22-06 11/8/83	To prevent possible loss of an aileron hinge pin	2/10/96	N/A per 3/8"	One		Signature: 
AD 86-24-07 1/7/87	To prevent engine power interruption due to loss of attachment of the engine controls	11/20/87 2/10/96	pk/w	One		Signature: 
AD 87-20-03 R2 9/24/90	To assure proper engagement of the seat locking mechanism and to preclude inadvertent seat	2/10/96	ck/w by inspection ref. seat rails	Recur iack		Signature: 
SEB 97-9A SEB 96-17 R1A	fuel strainer assy proton pins	12/1/97 12/1/97	N/A per no changes N/A per overhaul dates			Signature: 

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Airworthiness Directive Compliance Record				Profile ID:	N75908
Date:	02/01/1996				
<u>Manufacturer</u>	<u>Model</u>			<u>Part #:</u>	
Textron Lycoming	O-320-H2AD			<u>Serial #:</u>	L1471-76

AD Number Ctry - AD# Rev - Para	1. Subject 2. Comments	Complied 1. Date 2. Time	1. Status 2. Method of Compliance/ Applicability	One or Recur	Next Due 1. Date 2. Time	1. Authorized by 2. Facility 3. Cert. Type 4. Cert. Number
AD 55-02-02 2/28/55	Engine accessory drive adapter gasket	2/10/96	N/A per model #	One		Signature: [Signature]
AD 59-10-07 7/1/59	Cylinder baffle clamps	2/10/96	N/A per model #	One		Signature: [Signature]
AD 63-23-02 8/6/68	Exhaust valves and valve guides	2/10/96	N/A per model #	Recur		Signature: [Signature]
AD 64-16-05 7/10/64	To preclude the possibility of engine oil being drained overboard as a result of oil seal	2/10/96	N/A per model	One		Signature: [Signature]
AD 66-20-04 8/27/66	To prevent further failures of oil filter adapter gasket, p/n 74904	2/10/96	N/A per model S/N	One		Signature: [Signature]
AD 73-23-01 1/13/77	To prevent piston pin failures resulting from grinding cracks which occurred during manufa	2/10/96	N/A per model	One		Signature: [Signature]


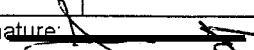



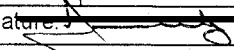
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Textron Lycoming	O-320-H2AD	<u>Serial #:</u> L1471-76

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AD 75-08-09 8/18/77	To prevent oil pump failures, inspect, replace and assemble the oil pump drive shaft and d	2/10/96	N/A per model	One		Signature: 
AD 77-07-07 2/20/79	To prevent the incorrect installation of the p/n lw-14784 or the p/n lw-15481 oil level ga	11/4/77 2/10/96	Plc/w per S.L 76-28	One		Signature: 
AD 77-20-07 8/31/78	To prevent hazards in flight associated with loose rocker arm retaining studs & failure of	2/10/96	C/w lifers replaced	One		Signature: 
AD 78-12-08 6/20/78	To prevent oil pump failures	4/14/78 2/10/96	N/A per S/L of engine	One		Signature: 
AD 78-12-09 2/20/79	To prevent possible crankshaft gear failure, remove and replace the crankshaft assembly, c	2/10/96	N/A due to engine S/L	One		Signature: 
AD 79-10-03 R2 7/1/80	To prevent loss of integrity of the engine to aircraft mounting bracket attaching bolts, c	2/10/96	C/w by Torquing bolt	One		Signature: 

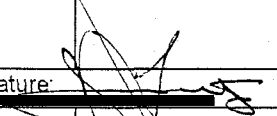

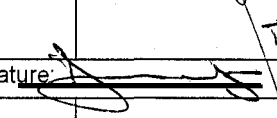
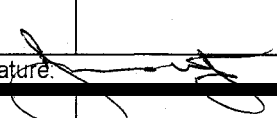
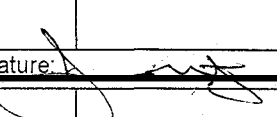
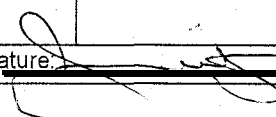
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Textron Lycoming	O-320-H2AD	<u>Serial #:</u> L1471-76

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AD 80-04-03 R2 4/25/88	To prevent hazards in flight associated with bent push rods on model o-320-h series engine	11/13/80 2/10/96 Rev. 15.97	c/w by adding oil additive Tact 3197.81 c/w per oil add	Recur every oil change		Signature: 
AD 80-14-07 7/7/80	To prevent failure of valves due to installation of improperly hardened upper exhaust valv	2/10/96	D/c/w by part A of AD 80-04-03	One		Signature: 
AD 81-18-04 R2 6/7/82	To prevent failure of engine oil pumps	2/10/96	Done @ 2000 hrs since	One		Signature: 
AD 87-10-06 R1 9/1/89	To prevent possible rocker arm failure and loss of engine power, inspect and rework or rep	2/13/96	N/A per model or S/W	One		Signature: 
AD 90-04-06 R1 5/28/91	To prevent oil line fracture and loss of engine oil	2/13/96	N/A per model or S/W	One		Signature: 
AD 91-14-22 8/19/91	To prevent loosening or failure of the crankshaft gear retaining bolt, which may cause sud	2/13/96	N/A per model or S/W	Recur		Signature: 

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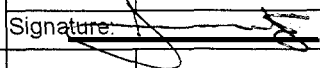
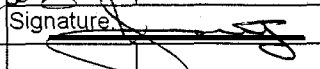
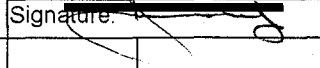
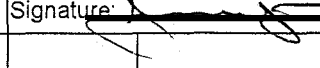
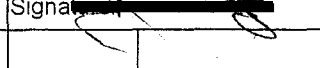

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Textron Lycoming	O-320-H2AD	<u>Serial #</u> : L1471-76

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AD 92-12-05 7/10/92	To prevent piston pin failure, or piston release, and engine failure	2/13/96	N/A per Model or S/N	One		Signature: [Signature] [Redacted]
AD 92-20-07 10/5/92	Superseded by 93-05-21	2/10/96	Superseded	One		Signature: [Signature] [Redacted]
AD 93-05-21 3/25/93	Superseded by 93-11-11	2/10/96	Superseded	One		Signature: [Signature] [Redacted]
AD 93-11-11 6/21/93	To prevent disruption of fuel flow to the engine, which can result in a loss of engine pow	2/13/96	N/A	One		Signature: [Signature] [Redacted]
AD 94-14-13 L 6/23/94 132 #04	To prevent detonation due to low octane, which can result in severe engine damage and subs	2/13/96	N/A By N#	One		Signature: [Signature] [Redacted]
AD 97-01-03	Cyl. head repair after Dec 15.95 by a repair facility	12/96	N/A by date of engine overhaul + any other cyl. work was prior	One		Signature: [Signature] [Redacted]

AD 97-01-13 Fuel hose repl 12/96 N/A hose was not replace
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 within the effective dates [Signature] [Redacted]

Report Produced By:	C.R.AVIATION, INC. Tamiami Airport	Address:	14592 S.W 129th. St Miami, FL 33186	Phone:	(305) 255-4222
Airworthiness Directive Compliance Record				Profile ID:	N75908
Date:	02/01/1996				
AD Number Ctry - AD# Rev - Para	1. Subject 2. Comments	Complied 1. Date 2. Time	1. Status 2. Method of Compliance/ Applicability	One or Recur	Next Due 1. Date 2. Time 1. Authorized by 2. Facility 3. Cert. Type 4. Cert. Number
<u>Manufacturer</u> Bendix Corporation	<u>Appliance Category</u> Magnetos	<u>Model</u> D-2000 SERIES		<u>Part #:</u> <u>Serial #:</u>	
AD 77-17-07 1/1/77	Superseded by 78-18-04	2/10/96	c/w per S/B 587	One	Signature: 
AD 78-09-07 R3 1/17/83	To preclude magneto or engine failure resulting from magneto impulse coupling failure	11/24/82 2/10/96 6/30/96 4/15/97	c/w per inspection Tech 2506. c/w per inspect Tech 3001.7 c/w per inspect	Recur any 500	Signature: 
AD 78-18-04 9/9/78	To preclude loss of ignition	2/10/96	P/c/w.	One	Signature: 
AD 79-12-07 6/19/79	To detect loose distributor block bushings	2/10/96	N/A by part #	Recur	Signature: 
AD 79-18-06 10/31/79	Ignition loss	2/10/96	P/c/w per mod ck at	One	Signature: 
AD 80-17-14 8/21/80	Precludes loss of ignition	2/27/8	P/c/w	One	Signature: 

<p>AD 82-11-05 6/9/82</p>	<p>Engine power loss and engine damage resulting from from looseness of the distributor gear</p>	<p>11/24/82 2/10/90</p>	<p>3/c/w @ oshl.</p>	<p>Rec'd W/A</p>	<p></p>	<p>AP [REDACTED] Signature [REDACTED]</p>
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